



## ELECTRONIC INSTRUMENT SYSTEM PRESENTATION ELECTRONIC INSTRUMENT **C** MENU System presentation



In comparison to old technology aircraft, the flight deck on the A320 is designed to be a comfortable uncluttered environment in which to work. By utilizing modern electronic display units, the presentation of information to the pilots has been improved.

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The Electronic Instrument System, (EIS), consists of six identical full color cathode ray tube display units. The EIS is divided into two subsystems:

- the Electronic Flight Instrument System, (EFIS), for which each pilot has two displays,
- the Electronic Centralized Aircraft Monitoring system, (ECAM), which uses the two displays in the center to provide information on the aircraft systems.

Let's look at the EFIS system first.





Flight parameters are displayed on Primary Flight Displays (PFD) while Navigation data is displayed on Navigation Displays (ND).

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#### **ELECTRONIC FLIGHT INSTRUMENT SYSTEM**









The two displays in the center are dedicated to the Electronic Centralized Aircraft Monitoring system (ECAM). At this stage we will simply introduce the ECAM displays and associated controls. In later modules, the ECAM system will be studied in detail.

## **ELECTRONIC CENTRALIZED AIRCRAFT MONITORING**



| ENGINE        |         |
|---------------|---------|
|               | VIB(N1) |
| 1550 KG 1540  | 0.2 0.2 |
| 0.71          | VIB(N2) |
| 17.2 LTR 17.2 | 0.2 0.2 |

| <u>AIR</u> | ∆P 1:5 P | LDG ELE | VAUTO 200 FT  |  |
|------------|----------|---------|---------------|--|
|            |          |         | CAB V/S FT/MN |  |
| CKPT       | FWD °C   | AFT     | 100           |  |
| 20         | 21       | 20      | CAB ALTET     |  |
|            | 20       | 21      | 5000          |  |
| TAT +      | 15 °C    |         | GW 69000 KG   |  |
| SAT +      | 15 °C    | 13 H 28 |               |  |

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The upper ECAM display is known as the Engine Warning Display (E/WD). The lower ECAM display is known as the System Display (SD). It is on this screen that various aircraft system parameters can be viewed by the pilots. As an example, we will cycle through the system pages for you.

### **ELECTRONIC CENTRALIZED AIRCRAFT MONITORING**



|                            | VIB(N1)              |
|----------------------------|----------------------|
| 1550 FUSED<br>1550 KG 1540 | 0.2 <sup>T</sup> 0.2 |
| OTI                        | VIB(N2)              |
| 17.2 LTR 17.2              | 0.2 0.2              |

| <u> Air c</u> | ∆P 1.5 F | LDG ELE<br>PSI | ev auto 200 ft |  |
|---------------|----------|----------------|----------------|--|
| <u> </u>      |          |                | CAB V/SFT/MN   |  |
| CKPT          | FWD °    | C AFT          | 100 8          |  |
| 20            | 21       | 20             | CAB ALT FT     |  |
| 20            | 20       |                | 5000           |  |
| TAT +         | 15 °C    |                | GW 69000 KG    |  |
| SAT +         | 15 °C    | 13 H 28        | 3              |  |

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System Display

System presentation

The presentation of system information is based on a "Need to Know" philosophy. This means that only the system information relevant to the particular phase of flight is presented to the pilots. You will see this demonstrated in the Normal and Abnormal operation modules.

#### **ELECTRONIC CENTRALIZED AIRCRAFT MONITORING**



| ENGINE           |         |
|------------------|---------|
| FUSED 1540       | VIB(N1) |
| 1550 - KG - 1540 | VIB(N2) |
| 17.2 LTR 17.2    | 0.2 0.2 |

|          | ∆P 1.5 P | LDG ELEV | AUTO 200 FT  |  |
|----------|----------|----------|--------------|--|
| <u> </u> |          |          | CAB V/SFT/MN |  |
| CKPT     | FWD 📍    | AFT      | 100          |  |
| 20       | 21       | 20       | CAB ALTET    |  |
| 20       |          | 21       | 5000         |  |
| TAT +    | 15 °C    | S.       | GW 69000 KG  |  |
| SAT +    | 15 °C    | 13 H 28  |              |  |

Engine/Warning Display

> System Display

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System presentation

Another philosophy that is used on the Flight Deck is the "Lights Out" principle. What this means is that when the aircraft is in its normal flight state there will be no white lights illuminated in any of the switches on the overhead panel. As an example, just before take off if you look up at the overhead panel, there should be no white lights illuminated on any of the switches. 

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Let's look at some switches and show the different possibilities.

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#### System deactivated

System activated

**ABNORMAL** Fault indication

For the majority of the switches on the overhead panel, the push button switch logic is;

Normal operation configuration - light out position

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- Normal operation configuration No light showing, (the LIGHTS OUT philosophy).
- Abnormal condition Amber FAULT light (this assists identification of the switch associated with an abnormal condition).
- Non lights out switch position White light. If normally the system should be operating and is deactivated a white OFF light is illuminated. If normally the system should not be operating and is activated a white ON light is illuminated.

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FAULT



Applicable system status

Temporarily selected switch

There are some switches on the overhead panel which are used on a  $\P$  temporary basis or may have an indication of their state. The logic is;

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- Temporary selection for operational reasons Blue ON light e.g. anti-ice
- Applicable system status Green light e.g. APU available. You will see this philosophy demonstrated throughout the course.











System presentation

Just below the ECAM screens, on the pedestal, is a switching panel for use in abnormal situations to restore data to the EFIS and ECAM displays. You will use this panel in the EIS Reconfiguration module. SWITCHING AIR DATA EIS DMC ECAM / ND XFR ATT HDG NORM NORM NORM NORM F/0 CAPT F/O CAPT F/O CAPT F/O CAPT R System presentation **MENU** 15/32

In front of each pilot there are two attention getters, a red MASTER WARNING, and an amber MASTER CAUTION. As a further means of getting the pilots attention, there is a loudspeaker on each side of the cockpit for aural alerts and voice messages.

Note; The loudspeakers can also be used to listen to ATC and the intercom.

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Now, let's go back to the EFIS system. For the EFIS displays, data from the Air Data and Inertial Reference System (ADIRS) plus Navigation data from the Flight Management and Guidance System (FMGS) is fed directly to three Display Management Computers (DMC).









The three identical DMC process the data and generate the images to be displayed.

**Under Normal circumstances:** 

- DMC1 supplies EFIS information to the Captains' PFD and ND
- DMC2 supplies the First Officers' PFD and ND.
- DMC3 is available as a backup. You will see the use of DMC3 in the abnormal operation module.







Now, let's look at the other EIS sub-system, ECAM, and how the ECAM displays get their data.

Sensors are fitted throughout the aircraft to monitor the various systems, including system controls operated in the flight deck.

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#### **AIRCRAFT SYSTEM SENSORS**

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Data for certain parameters, for example Fuel Quantity and Engine Primary Indications, is routed directly from the system sensors to the , three DMC. **(**) Note that there are separate channels within each DMC for ECAM and EFIS. SE.



System presentation





For the majority of the systems the sensors supply data to 2 System Data Acquisition Concentrators (SDAC). The SDAC acquire system data, process it, and send System page data to the 3 DMC.



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# The ECAM System



All the components shown can be collectively called the ECAM system.

We will study the use of the ECAM system in a separate module.



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In addition to EFIS and ECAM, time measurement devices are provided.

The master time reference for all aircraft systems is provided by a clock, located on the right lower side of the main panel.

In this example the time is 1328.





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In this module we have introduced you to the Electronic Instrument System, (with its two sub sections EFIS and ECAM), and the clock. In the next modules we will concentrate mainly on the ECAM system and then later in the course return to look at the EFIS displays in greater detail.

## Module Complete

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<u>System</u> presentation

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| <b>EXAMPLE 1 INTEGRATE</b> |               |      |  |
|----------------------------|---------------|------|--|
| EFIS DISPLAYS              |               |      |  |
| ECAN                       | ECAM DISPLAYS |      |  |
| SYSTEM ARCHITECTURE        |               |      |  |
| CLOCK                      |               |      |  |
|                            |               | _    |  |
| AUDIO                      | GLOSSARY      | FCOM |  |
| RETURN                     |               | EXIT |  |

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